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10AE73

Seventh Semester B.E. Degree Examination, Dec.2017/Jan.2018
Aircraft Stability and Control

Time: 3 hrs.

Max. Marks:100

Note: Answer any FIVE full questions, selecting at least TWO questions from each part.

PART – A

- 1 a. With neat sketch explain various aerodynamic forces on an airfoil section. (06 Marks)
 b. Describe and explain with necessary equations contribution of airframe components. (14 Marks)
- 2 a. Derive an expression for elevator effectiveness. (08 Marks)
 b. Derive and explain elevator required for landing. (12 Marks)
- 3 a. Obtain an expression for (δ_c) free elevator deflection condition for stick free condition. (06 Marks)
 b. Define stick free neutral and static free static margin and obtain relevant expression. (14 Marks)
- 4 a. Define static directional stability of aircraft and criteria with neat sketches and explain. (05 Marks)
 b. Explain : i) Adverse Yaw ii) Cross – wind landings iii) Asymmetric power iv) Spin recovery v) Rudder lock and the Dorsal fin. (15 Marks)

PART – B

- 5 a. Explain methods of aileron balancing. (12 Marks)
 b. Obtain relation for roll control power $C_{L\delta a}$. (08 Marks)
- 6 a. Define longitudinal dynamic stability of aircraft and plot types of modes of motion and discuss Phugoid and short period motion. (10 Marks)
 b. Develop governing equation for airplane in pure pitching motion and discuss A.O.A time history of pitching model for various developing ratios. (10 Marks)
- 7 a. Obtain the expression $\frac{P_{ss}b}{2U_0} = -\frac{C_{L\delta a}}{C_{Lp}} \cdot \Delta\delta a$. (12 Marks)
 d. Calculate the roll response of Tejas airplane $\tau, p_{ss}, L_{\delta a}$ to 5° step change in aileron deflection. Assume the airplane is flying at sea level velocity of 87 m/s. Consider the airplane configuration $C_{Lp} = -0.285\text{rad}^{-1}$, $C_{L\delta a} = 0.039\text{rad}^{-1}$, $S = 18 \text{ m}^2$, $b = 6.7 \text{ m}$. (08 Marks)
- 8 Write short notes on : (20 Marks)
 - a. Wing rock.
 - b. Roll control reversal.
 - c. Spring roll approximation.
 - d. Dutch roll approximation.

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